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# Investigation of the morphologies of chelate flame-sprayed metal oxide splats

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#### ABSTRACT

This paper reports an investigation of  $Er_2O_3$  splats formed by metal–ethylenediaminetetraacetic acid (EDTA) complex particles that were applied onto aluminum alloy substrates by flame spraying. The splat morphologies and coating microstructures were analyzed under different conditions. The effects of the in-flight particle temperature of the carrier gases and the impact particle deposition temperature on solidification of the molten droplets were evaluated with a rotating dodecahedron stage operated at velocities of 30, 75, and 90 rpm and with different temperatures and angles of incidence. Various techniques were used to analyze the surfaces and cross sections of the splats and coatings formed under the different conditions. Most of the splats underwent transitions from disk shapes (ratio *>* 85 %) to irregular shapes (circularity of 0.61–0.80) with increasing rotational velocity. The spray process provided a coating with a porosity of 23.3 % at a powder flow rate of 20 g/min. The temperature reached during the chelate flame spraying (CFS) process and the condensing velocity of the deposited splat, rather than the in-flight particle temperature, were found to control the splat morphology. This synergistic condensation effect paves the way to successful deposition of high-temperature structural ceramic coatings such as  $Er<sub>2</sub>O<sub>3</sub>$  with the CFS method.

### **1. Introduction**

Sprayed airborne particles undergo heating and acceleration after injection into the thermal plume during applications of thermal spray coatings [\[1\].](#page-10-0) The molten (or semimolten) droplets are propelled toward the substrate at high speeds. On impact, the droplets form splats, i.e., they rapidly spread and solidify on the substrate. During deposition, the buildup of these individual splats leads to the formation of coatings (or films) with the required surface characteristics. Therefore, splats with various morphologies and shapes provide the foundation of the coating [2–[5\]](#page-10-0). The morphologies and shapes of the deposited splats depend on multiple factors: the high-temperature thermal-physical properties of the powder material (especially ceramics, with melting points *>* 2100 ◦C), the substrate temperature, and the droplet solidification

characteristics [\[1\]](#page-10-0). A number of studies have been focused on contact by droplets sprayed perpendicular to the substrate surface and neglected the angle of incidence [\[6](#page-10-0)–9]. Most microstructural studies on the effects of these parameters on the splat morphology have involved particle deposition by plasma spraying due to its wide range of temperatures (typically from 8700  $°C$  to 12,000  $°C$ ) [\[10,11\]](#page-10-0); thus, many powder materials are fully melted. Generally, the coatings formed by the plasma spray processes are dense and exhibit lower porosity (porosity *<*1 %) than those deposited by combustion processes  $[12-14]$ . However, these qualities make it difficult to synthesize any required coatings or microstructures, such as porous coatings (porosity*>*30 %), and the high energy consumption and cumbersome processes further limit the application of these methods  $[15,16]$ . Thus, the development of a new spraying method for depositing metal-oxide coatings that provides

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<span id="page-1-0"></span>

**Fig. 1.** (a) Model image for the synthesis of  $M_2O_3$  by the chelate flame spraying process. (b) The chemical reaction of EDTA⋅M⋅H.

stability, such as thermal corrosion resistance, is still a great challenge.

A mixture of  $H_2-O_2$  or  $C_2H_2-O_2$  is commonly used as the combustion gas in flame spraying, which restricts the flame stream to relatively moderate temperatures (typically approximately 2300 ◦C) [\[17\]](#page-10-0). The restricted temperatures make this deposition method inapplicable to ceramic coatings prepared by high-temperature melting, such as yttrium oxide [\[18,19\].](#page-10-0) Therefore, researchers have proposed a new spray process using a metal–ethylenediaminetetraacetic acid (EDTA) complex [\[1,20,21\]](#page-10-0). In this process, the metal–EDTA powders are introduced to the substrate by a flame-spraying method, so we refer to it as chelate flame spraying (CFS) [\[22\]](#page-10-0). During the CFS process, the complexed powders are placed into a feed unit (5 g/min-15 g/min) and transported by a flowing carrier gas (air,  $N_2$  or  $O_2$ ) into the flame plume [\[23\]](#page-10-0). These metal–EDTA powders decompose and oxidize in the flame and form metal oxide particles, which are subsequently deposited on the substrate. However, the lower surface temperature of the substrate in the chelate flame spraying process results from the  $H_2O$  (water) generated in (a) the reaction of  $H_2$  and  $O_2$  in the  $H_2-O_2$  flame and (b) the decomposition reactions of the raw materials (EDTA⋅M⋅H, M: metal), as shown in Fig. 1. Moreover, acetylene burns too early and is not hot enough, the precursors cannot completely decompose the raw materials (EDTA), and the  $H_2$ -O<sub>2</sub> flame is more conducive to the preparation of clean coatings with no carbon residues. Therefore, it is of great significance to systematically study the morphology of a coating on a low melting substrate, such as an aluminum alloy, by the CFS method. The splat deposition process fundamentally consists of particles and droplets impacting the substrate and then spreading and solidifying [\[24\]](#page-10-0). The present study investigates the influence of the preparation parameters on the coating, such as a powder flow rate of  $5$  g/min and an  $O_2$  carrier gas with a flow rate of 7.1 L/min, which are used to deposit dense coatings with low porosity *<*3 % [\[1\].](#page-10-0) The above results mean that the splat particles can be changed to design the structure of the oxide ceramic coating  $(Y_2O_3$  and  $Er_2O_3$ ) by controlling the decomposition mechanism of EDTA; apparently, EDTA works better, and the flame spraying method overcomes the difficulty of preparing coatings with high melting points because the combustion and decomposition processes of the EDTA complex are exothermic [\[23\]](#page-10-0). In previous studies, we used the CFS method to synthesize ceramic materials (Y<sub>2</sub>O<sub>3</sub> and Er<sub>2</sub>O<sub>3</sub>)

and deposited coatings with cross-sectional porosities of 2–33 % on different substrates, such as stainless steel, aluminum alloys and quartz [24–[26\]](#page-10-0). Furthermore, droplet spreading played a key role in determining the microstructure, including the shapes and sizes of the pores and gaps between the splats of the deposited coatings. In previous studies, the cost of a CFS-sprayed  $Y_2O_3$  coating was similar to or greater than that of a thermal barrier coating synthesized with an atmospheric plasma spray (APS) method  $[25,27,28]$ . Additionally, deposited Y<sub>2</sub>O<sub>3</sub> coatings exhibited strong adhesion and good thermal shock resistance on aluminum alloy substrates [\[29,30\].](#page-10-0) Moreover, there was evidence that the splat morphologies were correlated with coating density, coating deposition efficiency and pores and oxide inclusions. Choosing the ideal carrier gas  $(N_2)$  and spray distance was shown to be an efficient and cost-effective way to alter the properties of the splats from fragmented to more desirable disk morphologies [\[22\]](#page-10-0). Other behaviors in CFS, such as the particle impact styles, are still not fully understood. Thus, there is still a need to establish quantitative correlations between splat morphologies and shapes and coating microstructures to achieve optimal designs.

Few studies have described the development of splat morphology and microstructure by using CFS with the process parameters used in this study. Herein,  $Er_2O_3$  splats and coatings were deposited by CFS onto aluminum alloy substrates. The impact angles of the airborne particles and droplets were set with a rotation apparatus (dodecahedral pattern) with rotational velocities of 30, 75 and 90 rpm. The overarching aim of this study was to investigate the spreading behavior of EDTA⋅Er⋅H raw materials with different cooling modes (fixation and rotation) and spraying parameters. To assess the circularity, the proportion of disklike splats, the splat particle sizes and the microstructures of the  $Er<sub>2</sub>O<sub>3</sub>$ splats and thick coatings, and the cross-sectional splat-substrate features were analyzed with scanning electron microscopy (SEM) and a watershed algorithm. We explained the effects of the CFS process parameters, including impact types and airborne particle temperatures, on the characteristics of the ceramic coating.

#### **2. Experimental procedure**

The powder feedstocks for the metal–EDTA complex, EDTA⋅Er⋅H (Chubu Chelest *Co*., Ltd.), were flame sprayed onto a polished aluminum alloy (A5052, 50  $\times$  50  $\times$  5 mm<sup>3</sup>) substrate. The chelate materials were synthesized with the following method. First, a stoichiometric amount of ethylenediaminetetraacetic acid disodium salt (EDTA-2Na, 99 %) was dissolved in 60 mL of D.I. water under constant magnetic stirring at room temperature. The concentration of the original solution was 0.1 M. To this solution, 1 M NaOH solution was added dropwise to maintain a pH of 11.5–13, which was monitored with a pH meter (METTLER-TOLEDO–PE20K), and this solution was labelled A. Afterward, an equivalent molar ratio of erbium nitrate hexahydrate  $(Er(NO<sub>3</sub>)<sub>3</sub>$ .6H<sub>2</sub>O, 99 %) was dissolved in 50 mL of D.I. water under constant magnetic stirring, a transparent solution formed and the solution was allowed to stir for an additional half an hour. Then, the transparent solution was mixed with solution (A) and placed at room temperature for 8–14 h. A



**Fig. 2.** (a) Powder state of the raw material EDTA⋅Er⋅H, (b) particle size distribution of the unscreened powder, and (c) particle size distribution of the 45 μm material.

<span id="page-2-0"></span>

**Fig. 3.** Schematic of the coating deposition apparatus with a rotating stage.

white precipitate was formed. The solid white precipitate was collected and washed with D.I. water and ethanol several times and oven-dried at 90  $\degree$ C for 5–8 h. The obtained sample was ground to form a fine powder.

The substrate surface was cleaned twice with acetone. The experimental setup used to deposit these splats can be described in a previous study [\[22\]](#page-10-0). We used 45-μm raw materials and unscreened EDTA⋅Er⋅H powder to perform the splat tests, as shown in [Fig. 2](#page-1-0) (a). In addition, to study the relationship between substrate temperature and particle morphology, we used the aforementioned flame spray equipment and a 12-sided rotating stage on which the substrate was fixed; the substrate temperature was reduced by rotation, as shown in Fig. 3.

Tables 1 and 2 list the parameters (powder flow rate, rotational velocity and velocity of the substrate) for the carrier gases ( $O_2$  and  $N_2$ ) that were used to deposit the 45-μm raw materials and unscreened EDTA⋅Er⋅H powder on the individual Er2O3 splats; the particle sizes are shown in [Fig. 2](#page-1-0) (b) and (c). To study the relationships between the individual splat morphologies and the coating structures,  $Er<sub>2</sub>O<sub>3</sub>$  coatings were deposited via a CFS system with a rolling velocity of 90 rpm; the spraying parameters are listed in [Table 3](#page-3-0).

The phases of the various synthesized coatings were characterized by X-ray diffraction (XRD, M03XHF22, Japan) with Cu-Kα radiation over a 2θ range of 10–90◦. A range of microscopic techniques, including fieldemission scanning electron microscopy (FE-SEM), was employed to observe the surface morphologies and cross sections of the splats and coatings. To study the morphologies of the splats, the SEM images were binarized with ImageJ to display the A5052 substrate and the splats with different colors, such as white and black. Furthermore, the overlapped splats in the binarized surface images were separated with the watershed algorithm. Afterward, the various morphological parameters of the splats were calculated from the surface images [\[31\]](#page-10-0); this technique was applied in our previous studies [\[22\].](#page-10-0) Important information about the splats was estimated from many surface images to obtain accurate values. The circularity was calculated with Eq. (1):

$$
Circularity = \frac{4\pi A}{p^2}
$$
 (1)

where A is the area of the splat and P is the perimeter of the splat. Then, a boundary tracking algorithm was used to determine the boundaries from the image contours obtained by binarizing the P regions with ImageJ software. In addition, when the shape of a splat was similar to that of a disk, the primary slat circularity approached one, and distorted splats had low circularities [\[22\].](#page-10-0) SmileView and ImageJ software were used for quantitative analyses of the  $Er<sub>2</sub>O<sub>3</sub>$  coating microstructures.

#### **3. Results and discussion**

The use of  $O<sub>2</sub>$  as the carrier gas not only increased the flame temperature but also facilitated rapid decomposition and oxidation of the chelate powder. The temperature of the in-flight particles was approximately 2400  $\degree$ C, which was higher than that produced by using N<sub>2</sub> as the carrier gas [\[23\].](#page-10-0) The raw materials of metal-EDTA complexes were placed in a feed unit and transported by the flowing carrier gas  $(N_2 \text{ or } N_1)$  $O_2$ ) to the spray gun. The gas–solid mixture was introduced into a  $H_2-O_2$ flame and reacted with oxygen after thermal decomposition of the EDTA in the CFS process. Moreover, in the chemical reaction of the metal-–EDTA complex powder in the flame, the EDTA⋅Er⋅H mixture reacted with  $O_2$  to form a metal oxide film, as shown in Eq.  $(2)$ :

$$
2EDTA\cdot Er\cdot H + 24O_2 \rightarrow Er_2O_3 + 20CO_2 + 13H_2O + 4NO_2
$$
\n
$$
(2)
$$

Therefore, when the carrier gas was changed from  $O_2$  to  $N_2$ , the reaction was inhibited, the thermal energy of the in-flight particles decreased, and the particle temperature decreased. These changes in the temperature of the impacting particles are expected to exert the greatest effect on droplet spreading. Additionally, the jet entrained large amounts of air, and the reaction still occurred in the  $N_2$  carrier gas.

#### *3.1. Changes in the splats with different powder flow rates*

[Fig. 4](#page-4-0) (a–d) shows the typical morphologies of the  $Er<sub>2</sub>O<sub>3</sub>$  splats produced at powder feed rates of 5, 10, 15 and 20 g/min when the carrier gas was  $O_2$ . There were no significant macroscopic differences in the splat morphologies due to the different powder feed rates. When  $N_2$ was used as the carrier gas, the numbers and sizes of the splats decreased as the feed rate of the powder was increased, as shown in [Fig. 4](#page-4-0) (h–k). In contrast, when different gases were combined with the  $O_2$  and  $N_2$ , there were significant differences in the morphologies of the particles and the number of splats. [Table 4](#page-5-0) shows the temperatures seen for the airborne particles at powder feed rates of 10 and 20 g/min with  $O_2$  as the carrier gas. As the powder feed rate was increased from 10 to 20 g/min, the airborne particle temperature and velocity decreased. This was explained with Eq.  $(3)$ , which was provided by F. Fanicchia et al.  $[32]$ ; as the powder feed rate increased with  $M_{\text{powder}}$ , the loading effect increased. The increased loading effect was why the airborne particle velocity and temperature were determined with this method. Furthermore, the effect of the parameters for the airborne particles on particle melting is shown in Eqs. (3)–(6) [\[32,33\]](#page-10-0):

$$
\int_0^{\tau} Q_{int} \cdot dt > Q_M \tag{3}
$$

**Table 1** 

Experimental conditions used to prepare splats via the deposition of EDTA⋅Er⋅H with different powder flow rates. O<sub>2</sub> was used as the carrier gas for samples (a)–(d), and  $N_2$  was used as the carrier gas for samples (h)–(k).

No.	Method	Raw material	Flow rate of powder $(g/min)$	Flow rate of carrier gas $(L/min)$	Distance (mm)	Number of scans	Carrier gas
a/h b/i c/i d/k	Fixed	Unscreened	5.0 10.0 15.0 20.0	7.1	150		$O_2/N_2$

<span id="page-3-0"></span>

**Table 3**  Parameters for the  $Er<sub>2</sub>O<sub>3</sub>$  coating depositions.

Sample	Flow rate of powder $(g/$ min)	Carrier gas type	Rolling velocity (rpm)	Distance (mm)	Flow rate of carrier gas (L/min)
(a) (b)	20 20	O <sub>2</sub> N <sub>2</sub>	90	150	7.1

$$
\tau = \frac{SOD}{V_p} \tag{4}
$$

$$
Q_{int} = h \left(\pi d_p^2\right) \left(T_{\infty} - T_p\right) - \left(\pi d_p^2\right) \epsilon \sigma_s \left(T_p^4 - T_a^4\right) [W] \tag{5}
$$

$$
Q_M = \frac{4}{3}\pi \rho_p d_p{}^3 (C_p (T_m - T_0) + L_m) [J]
$$
 (6)

where  $Q_{int}$  is the heat transferred to the particles during flight;  $Q_M$  is the heat required to melt the particles;  $\tau$  is the in-flight dwell time of the particles; SOD is the stand-off distance;  $V_p$  is the particle velocity;  $d_p$ ,  $\rho_p$ , and C*p* are the particle diameter, density and specific heat, respectively; *h* is the convective heat transfer coefficient;  $T_{\infty}$  is the flame temperature at the particle surface; *ε* is the particle emissivity;  $σ<sub>s</sub>$  is the Stefan–Boltzmann constant;  $T_p$  and  $T_\alpha$  are the temperatures of the particle and surroundings, respectively; and  $T_m$ ,  $T_0$  and  $L_m$  are the particle melting temperature, initial temperature and latent heat of fusion, respectively. In Eqs. [\(3\)](#page-2-0)–(6), intraparticle thermal conduction and oxidative behavior were neglected. Thus, when Q*int* was higher than Q*M*, the particles in the flame melted. These equations suggested that as the powder feed rate was increased, the loading effect would increase, and the temperature and velocity of the airborne particles would decrease. Moreover, a lower velocity led to a longer heating time and higher particle temperatures (unless the standoff distance was too large so that excessive cooling occurred) in the traditional spray flame. However, these effects decreased the amount of heat transferred (Qint) (due to cooling of the flame by the higher feed rate) in spite of the lower particle velocity and longer residence time in the flame, which negatively affected splat melting and diffusion in the chelate spraying process, perhaps by affecting the EDTA decomposition rate. Thus, as the powder feed rate was increased, the temperature of the airborne particles was expected to decrease, and diffusion during the splat formation deteriorated. However, based on rapid splat solidification according to the thermal conductivity of the A5052 substrate, we expected that the proportions of splash-like particles, unmelted particles and resolidified splats would increase because the feed rate increased, which indicated that during the impact and deposition of particles, those particles that did not flatten were detached from the substrate.

Furthermore, the circularities and proportions of the disk-like splats and the splat sizes were measured from SEM images of the surface ([Fig. 4\)](#page-4-0), as listed in [Table 5.](#page-5-0) The use of  $N_2$  as the carrier gas increased the circularities and proportions of the disk-like splats with increasing powder feed rate. Moreover, splashing was likely if the solidification was not fast, which happens with warmer substrates or superheated melts, such as when  $O_2$  is used as the carrier gas (shown in [Fig. 4](#page-4-0)(a)). According to this analysis, the Mpowder and Q*int* values of the airborne particles increased, which resulted in a lower temperature for each particle. Additionally, the measurement results showed that when the powder flow rate was 15 g/min, the splats all exhibited higher disk-like ratios when  $N_2$  and  $O_2$  were used as the carrier gases. Thus, the temperatures were not sufficiently high for all of the airborne particles to reach the melting point and decompose the EDTA⋅Er⋅H. This was consistent with the results of our previous studies [\[1\]](#page-10-0). This meant that a powder feed rate of 15 g/min was the cutoff point in this study. Additionally, the adhesion rate decreased at a powder feed rate of 20 g/min. Therefore, it is reasonable to hypothesize that the powder feed rate of the EDTA⋅Er⋅H and carrier gas used in this study controlled the

<span id="page-4-0"></span>

Fig. 4. Images of the surfaces of Er<sub>2</sub>O<sub>3</sub> splats deposited on the A5052 substrate at different powder flow rates.; O<sub>2</sub> (a)/N<sub>2</sub> (h): 5 g/min, O<sub>2</sub> (b)/N<sub>2</sub> (i): 10 g/min, O<sub>2</sub> (c)/N<sub>2</sub> (j): 15 g/min, and O<sub>2</sub> (d)/N<sub>2</sub> (k): 20 g/min.

#### <span id="page-5-0"></span>**Table 4**

Temperature changes of the in-flight particles in the EDTA⋅Er⋅H powder with different spray distances and powder feed rates while using  $O_2$  as the carrier gas.



#### **Table 5**

Results for the samples in which  $Er<sub>2</sub>O<sub>3</sub>$  splats were deposited on the A5052 substrate with different powder flow rates.

Flow rate of powder $(g/min)$	Circularity $\sqrt{SD(-)}$	Disk-like ratio /SD (%)	Splat particle size $/SD$ ( $µm$ )
5.0	0.66/0.02	65.8/4.6	15.4/1.1
10.0	0.67/0.01	66.8/3.6	16.4/0.4
15.0	0.70/0.01	71.6/2.1	17.3/0.9
20.0	0.67/0.01	65.8/3.0	16.8/0.8
5.0	0.66/0.01	63.6/2.5	15.4/0.4
10.0	0.70/0.02	75.0/2.5	13.6/0.6
15.0	0.79/0.01	88.5/1.1	11.6/0.4
20.0	0.80/0.04	87.9/4.8	10.2/0.2

deposition of porous or dense splats.

#### *3.2. Changes in the splat state as a function of the rotational velocity*

[Fig. 5](#page-6-0) shows SEM images of the surfaces of the coatings sprayed on the A5052 substrate with  $O_2$  carrier gas (samples (e), (f), and (g)) and  $N_2$ carrier gas (samples (l), (m), and (n)). In a previous study, we used a contact thermometer to measure the temperature of the substrate after spraying at 579, 454, and 441 K for each rotational speed (30, 75, and 90 rpm, respectively) [\[25\]](#page-10-0).

In the SEM images showing the surface morphologies of the  $Er<sub>2</sub>O<sub>3</sub>$ splats, the types of deposited splats differed significantly (finger-shaped, splash-shaped, voided, microcracked, partially molten, and malformed splats). When  $O_2$  was the carrier gas, a shattered splatter deposition and splashing pattern upon impact was observed when the rotational velocity was 30 rpm, as shown in Fig.  $5$  (e). In addition, the splats were transformed from shattered splatters to disk-shaped splatters as the rotational velocity was increased from 75 to 90 rpm (shown in [Table 2\)](#page-3-0) due to the velocity of the substrate; this was much lower than the particle velocity, resulting in a faster cooling rate, as shown in Fig.  $5(f)$ , (g). However, when  $N_2$  was the carrier gas, the microstructures in the deposited disk-shaped splats were relatively uniform. In contrast to the use of  $O<sub>2</sub>$  as the carrier gas, the splat diameters of the deposits decreased as the rotation speed increased. Overlaps and microcracking were clearly observed in [Fig. 5](#page-6-0) (l). Moreover, ideal disk-shaped splats were observed on the A5052 substrate when a rotational velocity of 30 rpm was used, which was similar to the observations reported by Markus Mutter et al. [\[34\]](#page-10-0) for the use of the plasma spraying method. With a rotational velocity of 75 rpm, there was some molten material in the splats, as shown in Fig.  $5$  (f). The long fingers in the splat were clearly formed at high rotational velocities (90 rpm) and were observed in the SEM photographs shown in Fig.  $5$  (n). This could have been caused by molten particles impacting the substrate as it rotated at high speed and applied a rotational force on the deposited splats [\[35\]](#page-10-0). Notably, the majority of the splats on the substrates were formed from fully melted particles, and some splats were disk-shaped, which was similar to the morphologies of plasma-sprayed splats [\[36\].](#page-10-0) However, the splats resulting from freezing-induced splashing had different appearances than those that fragmented during impact (Fig.  $5$  (g) and (n)). Thus,

when the molten particles impacted the lower temperature substrate, instant coagulation occurred, and the number of irregular boundaries increased. For the above case, the deposited splat size was greater when the  $N_2$  carrier gas was used than when the  $O_2$  carrier gas was used. Additionally, it can be inferred from [Fig. 5](#page-6-0) (n) that the splat with larger fingers radiating out from its periphery was formed with a higher rotational velocity; additionally, spreading was considerably faster than solidification.

In addition, the surface images were used to calculate the degrees of circularity, the proportions of disk-like splats, and the splat sizes of the particles deposited at different rotational velocities with  $O_2$  and  $N_2$ carrier gases (Eq. [1](#page-2-0)); the results are shown in [Table 6](#page-6-0). According to the results of the calculations, the circularities of the slats and the proportions of disk-like slats decreased below 75 rpm for both the  $O_2$  and  $N_2$ carrier gases. In the case of 90 rpm rotation, these values increased. In comparison, the splat morphologies showed greater degrees of circularity and smaller disk-like splat particles when  $O_2$  rather than  $N_2$  was used the carrier gas. In addition, for both gases, as the number of rotations increased, the splat sizes decreased, and the largest splat diameters were 10–15 μm, as shown in [Fig. 6.](#page-7-0) The chart in [Fig. 7](#page-7-0) shows the circularities of the  $Er_2O_3$  splats deposited on the A5052 substrates at different rotational velocities. When  $O<sub>2</sub>$  was used as the carrier gas, the splats with circularities of 0.8 (from 0.7 to 0.79) were increased in abundance as the rotational velocity increased. The number of splats with circularities of 0.9 (from 0.8 to 0.89) was clearly increased at rotational velocities of 30 and 90 rpm. Furthermore, splats with circularities of 0.6 (from 0.5 to 0.59), 0.7 (from 0.6 to 0.69), and 0.9 (from 0.8 to 0.89) increased in number as the rotational velocity increased with  $N_2$ used as the carrier gas. Finally, when  $O<sub>2</sub>$  was the carrier gas, there were more splats with a greater circularity and a greater proportion of small splats compared to when  $N_2$  was the carrier gas. Furthermore, the distribution of the splat particle sizes was narrower when  $O_2$  was the carrier gas.

The effects of the rotational velocity (rpm) on the circularities and proportions of disk-like splats and the splat particle sizes are shown in [Fig. 8](#page-8-0). The circularities and proportions of disk-like splats were lower with a rotational velocity of 75 rpm; in contrast, these parameters were increased at 90 rpm for the  $O_2$  and  $N_2$  carrier gases. In addition, the splat particle sizes decreased with the rotational velocity, as shown in [Fig. 8](#page-8-0)  (c). This occurred because the rotational velocity increased as the substrate temperature decreased, and the results are shown in [Table 7](#page-8-0). When the rotational velocity was increased to reduce the scan time corresponding to the flame, the substrate temperature during deposition decreased. When the substrate temperature decreased, the temperature gradient between the splat and the substrate increased, and the effect of the cooling rate increased. Therefore, when the airborne particles impacted the substrate, they solidified or became partially solidified before they flattened, so the number of splashes or splats that were not flattened decreased. With the above conditions, the proportion of splats with splash-like shapes increased, and the circularity and the proportion of the disk-like splats decreased at a rotational velocity of 75 rpm. In addition, since the splat particle sizes decreased, the number of resolidified particles increased. At 90 rpm, the splat particle sizes continued to decrease, the numbers of resolidified particles and nonflat splats increased, and the circularities increased on the surface; thus, the circularity and proportion of disk-like splats also increased. Depending on the rotational velocity, this behavior might cause an increase in the coating porosity.

#### *3.3. Microstructure of the coating*

During deposition, the in-flight particles that fully absorbed heat while passing through the flame zone melted fully and were spread uniformly and completely upon impacting the substrate surface, which produced a smooth, flat, dense fully melted zone [\[37\]](#page-10-0). Thus, the majority of ceramic coatings with high melting points can be prepared by

<span id="page-6-0"></span>

Fig. 5. SEM images of the surfaces of Er<sub>2</sub>O<sub>3</sub> splats deposited on the A5052 substrate at different rotational velocities; O<sub>2</sub> (e)/N<sub>2</sub> (l): 30 rpm, O<sub>2</sub> (f)/N<sub>2</sub> (m): 75 rpm, and  $O_2$  (g)/ $N_2$  (n): 90 rpm.

**Table 6**  Results for samples in which the  $Er<sub>2</sub>O<sub>3</sub>$  splats were deposited on the A5052 substrate at different rotational velocities.

No. / Carrier gas	Rolling velocity (rpm)	Circularity $\sqrt{SD(-)}$	Disk-like ratio /SD (%)	Splat particle size /SD $(\mu m)$
(e)/O <sub>2</sub>	30	0.72/0.01 0.70/0.00	77.7/4.0	12.0/0.5
(f)/O <sub>2</sub>	75		74.3/3.5	10.4/0.4
(g)/O <sub>2</sub>	90	0.74/0.02	84.1/7.2	10.2/0.5
$(l)/N_2$	30	0.70/0.02	69.7/6.4	17.3/0.7
$(m)/N_2$	75	0.66/0.00	61.8/5.3	14.4/0.4
$(n)/N_2$	90	0.70/0.01	73.1/1.11	13.8/1.8

plasma spraying [\[36,38\].](#page-10-0) The combination of physical and chemical means, such as the pyrolysis of precursors in the early stage and oxidation in the middle stage, as well as regulation of the parameters related to the impact of flying particles on the substrate in the later stage. It will be more promising to realize the preparation and design of ceramic coatings such as those prepared by the EDTA method if the method is applied to lower melting temperature materials. The XRD patterns of the  $Er_2O_3$  coatings are shown in [Fig. 9.](#page-9-0) The peaks in the XRD profiles were assigned by using International Centre for Diffraction Data

(ICDD) cards as references. The deposited  $Er<sub>2</sub>O<sub>3</sub>$  coating exhibited a cubic crystalline phase (ICDD, No. 00–008-0050). The XRD spectra always contained peaks corresponding to the original phase of the coating used with  $N_2$  or  $O_2$  as the carrier gas at a rotation speed of 90 rpm, indicating that the coating did not differ from that obtained without the rotating stage [\[1\]](#page-10-0). In addition, the peak intensity of the coating phase hardly changed with changes in the spraying conditions.

To verify the influence of the splat morphologies described above on the coating microstructures, coatings were produced with the two carrier gases at a rotational velocity of 90 rpm and a powder flow rate of 20 g/min. [Fig. 10](#page-9-0) shows that the microstructures of the coatings were significantly different in these cases. Compared to the results obtained with  $O<sub>2</sub>$  as the carrier gas, more large pores and meshwork were observed in the coating made with  $N_2$  as the carrier gas, as shown in [Fig. 10](#page-9-0) (a) and (b). A rotation splat interfacial line was formed, and the line was marked, as shown at high magnification in [Fig. 10](#page-9-0) (a<sub>1</sub>). Moreover, the same microstructure can be observed in Fig.  $10(a<sub>2</sub>)$ . The irregular shapes of splats clearly overlapped to form large voids between the splats, as shown in Fig.  $10$  (b<sub>1</sub>) and (b<sub>2</sub>), which was consistent with the morphologies of the individual splats ( $Fig. 5(n)$ ). Additionally, with the decreases in substrate and in-flight particle temperatures, the proportion of meshwork structures increased, as shown in [Fig. 10](#page-9-0), which

<span id="page-7-0"></span>

Fig. 6. Particle sizes of the  $\text{Er}_2\text{O}_3$  splats deposited on the A5052 substrate at different rotational velocities: (a) 30 rpm, (b) 75 rpm, and (c) 90 rpm.

was roughly consistent with a previous study of the meshwork resulting from a coating deposition process [\[1\]](#page-10-0). Moreover, with the crosssectional images, we evaluated the thicknesses and porosities of the coatings. The sample in  $Fig. 10$  (b) had a porosity more than two times greater than the porosity of the sample in Fig.  $10$  (a), which exhibited a porosity of 23.3 %. The details are listed in [Table 8.](#page-9-0) Based on these results, it was evident that the microstructures of the coatings differed because of the lower in-flight particle temperatures. After impact, the droplets began to spread as a result of inertia due to the substrate rotation. In other words, after impinging the rotating substrate, the droplets quickly began to flatten and formed irregular shapes during solidification. This was accompanied by simultaneous losses of kinetic and thermal energies, which led to the differences observed for the solidified morphologies of the splats and built-up porous coatings. As mentioned above, airborne metal–EDTA particles that fully absorb heat decompose, oxidize, and melt thoroughly in CFS systems. Thus, the droplet impact styles play a significant role in defining the spreading behavior at the periphery of a splat. Therefore, it is reasonable to assume that the CFS method has promise for synthesizing high-quality structural ceramic coatings.

### **4. Conclusions**

In this study,  $Er<sub>2</sub>O<sub>3</sub>$  splats and thick coatings were prepared with CFS. The spreading behavior of a metal–EDTA complex mixed powder sprayed on an A5052 substrate was investigated. The following is a summary of the findings:

• The majority of the  $Er<sub>2</sub>O<sub>3</sub>$  splats were formed from fully melted particles and exhibited typical morphologies in both carrier gases, such as disk- and splash-shaped morphologies. Since relatively less splashing was observed when  $N_2$  was used as the carrier gas, the



Fig. 7. Circularities of the Er<sub>2</sub>O<sub>3</sub> splats deposited on the A5052 substrate at different rolling velocities: (a) 30 rpm, (b) 75 rpm, and (c) 90 rpm.

droplets prepared with  $O_2$  spread more easily and formed fragmented splats due to overmelting. However, the effect became less pronounced as the flow rate of the powder was increased from 5 to 20 g/min. The rotations decreased the substrate temperature and modified the splats formed by enhancing droplet melting during spreading.

• When  $O_2$  was used as the carrier gas with high-speed rotation (90 rpm), the  $Er<sub>2</sub>O<sub>3</sub>$  coating had a whirl-like lamellar structure and some porosity (10.3 %). When  $N_2$  was used as the carrier gas, the coating porosity increased (23.3 %) with increasing irregularity of the splat morphology because of the mode of solidification. Therefore, the morphologies and characteristics of the splats (such as the circularities and proportions of the disk-like splats) had direct effects on the coating microstructures deposited by CFS.

In summary, flower-like splats and disk-like splats were often formed by droplets with appropriate cooling velocities after impacting the substrate. Moreover, compared with traditional melted raw materials and parameter control methods, the CFS system leads to more complex coating deposition behavior, so more key factors, such as the decomposition speed of the metal–EDTA complex, must be explored to evaluate this flame spraying method completely.

### **CRediT authorship contribution statement**

Yanxin Dan: Conducting a research and investigation process; Preparation.

**Xiaomei Liu:** specifically critical review.

**Yu Wang:** Management activities to annotate (produce metadata);

<span id="page-8-0"></span>

Fig. 8. Results for Er<sub>2</sub>O<sub>3</sub> splats deposited on the A5052 substrate at different rotational velocities; (a) Splat circularity, (b) proportion of disk-like splats, and (c) splat particle size.





Application of statistical.

**Jing Huang:** Provision of study materials.

**Hidetoshi Saitoh:** Ideas.

**Yi Liu:** Acquisition of the financial support for the project leading to this publication.

Hua Li: Formulation or evolution of overarching research goals and aims.

<span id="page-9-0"></span>

**Fig. 9.** XRD patterns of Er<sub>2</sub>O<sub>3</sub> prepared by CFS at 90 rpm with different carrier gases; (a) O<sub>2</sub> as the carrier gas, and (b) N<sub>2</sub> as the carrier gas.



Fig. 10. SEM images showing the cross-sectional morphologies of Er<sub>2</sub>O<sub>3</sub> coatings prepared with high rotational velocities: (a)/(a<sub>1</sub>)/(a<sub>2</sub>) macromorphology/high magnification,  $(b)/(b_1)/(b_2)$  macromorphology/high magnification.

## **Table 8**

Estimated thicknesses and cross-sectional porosities of  $Er<sub>2</sub>O<sub>3</sub>$  coatings prepared with a rotational velocity of 90 rpm and a powder flow rate of 20 g/min.



# **Declaration of competing interest**

The authors declared that they have no conflicts of interest to this work.

We declare that we do not have any commercial or associative interest that represents a conflict of interest in connection with the work.

## **Data availability**

Data will be made available on request.

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